

The present invention relates to an attachment device, in particular a front attachment device, for attaching a turbo engine (1) to an aircraft pylon including an upper brace (12, 12') fitted with rigid fastening means to the pylon, a first (16) and a second (18) connecting rods, each connected to the engine, via a first ball joint connection (161, 181; 161', 181'), on the one hand, and to the upper brace (12, 12') via a second ball joint connection (165, 185; 165', 185'), on the other hand. The device is characterized in that each connecting rod (16, 18) includes a third stand-by link (167, 187; 176', 187'). The invention enables the realization of elements of the same shape. There is thus provided interchangeability between these elements and a reduction in the production and management costs. Besides, the risks of error during assembly are reduced.