

A cooled gas turbine engine blade comprising a foundry part and a longitudinal sleeve obtained by the forming of metal sheet, the foundry part including a longitudinal body provided with a longitudinal cavity with first and second apertures at its ends, the sleeve being mounted in the cavity being held therein by soldering or brazing to the wall of the first aperture, and an end portion of which is free to slide in the second aperture forming a slide, characterized by the fact that the end portion and the slide are in a sliding contact relatively to each other along surfaces formed by machining. According to the preferred embodiment a sealing insert is positioned between the end portion of the sleeve and the wall of the second aperture of the body of the blade. The sealing insert is soldered or brazed to the end portion of the sleeve.